



# MAXIMIZING SCIENCE RETURN: A REPRESENTATIVE TRAJECTORY for DYNAMO

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## ASSUMPTIONS

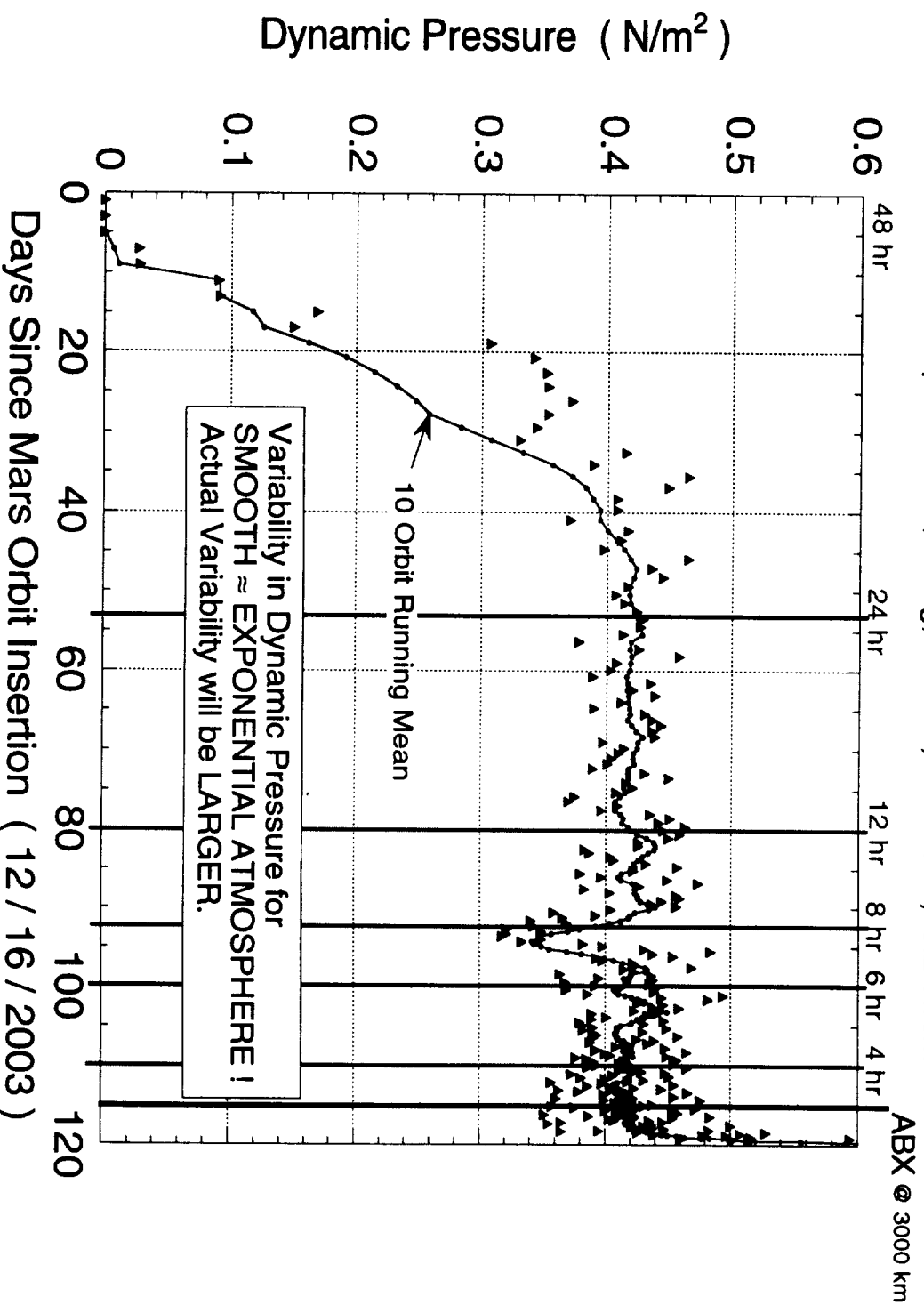
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- Launch Piggyback on Ariane V (Geosynchronous Transfer)
- Lunar and Earth Gravity Assist into Transfer Orbit.
- Propulsive Capture: 48 hour Orbit with 250 km Periapsis.
- SPACECRAFT
  - MASS = 80 kg (“Twin” Pallet Option)
  - AREA = 2.5 m<sup>2</sup> for Drag Pass (0.5 m<sup>2</sup> for Mapping)
  - $C_D = 2.0$
- Aerobraking Exit Maneuver (ABX) when Apo. = 3000 km.
- Try to keep periapsis as low as possible during mapping.

# DYNAMIC PRESSURE

## DYNAMO AEROBRAKING PHASE

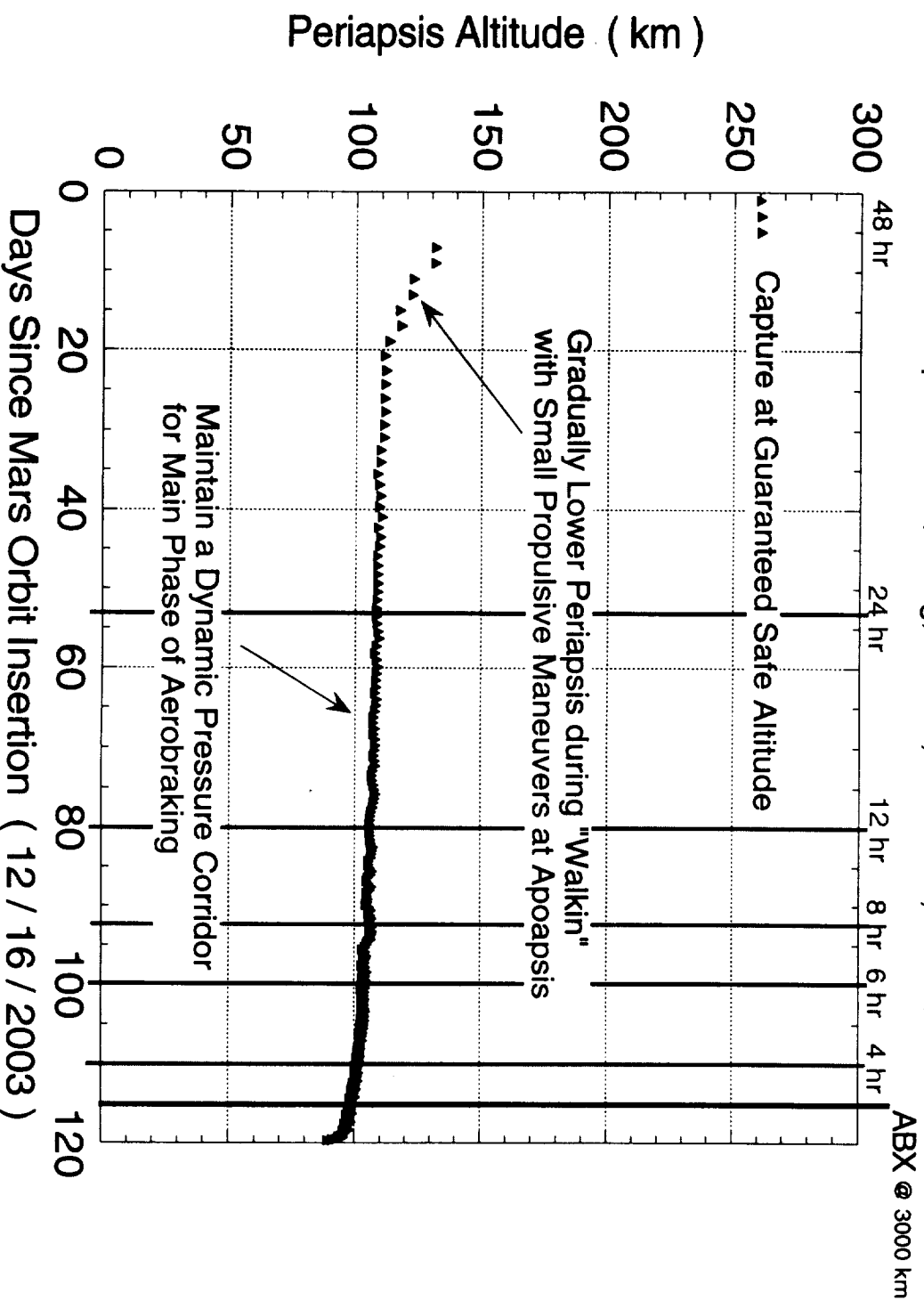
48 hr Capture Orbit, 80 kg, 2.5 m<sup>2</sup>, 0.4 N/m<sup>2</sup>, 0.2 W/cm<sup>2</sup>



# PERIAPSIS ALTITUDE

## DYNAMO AEROBRAKING PHASE

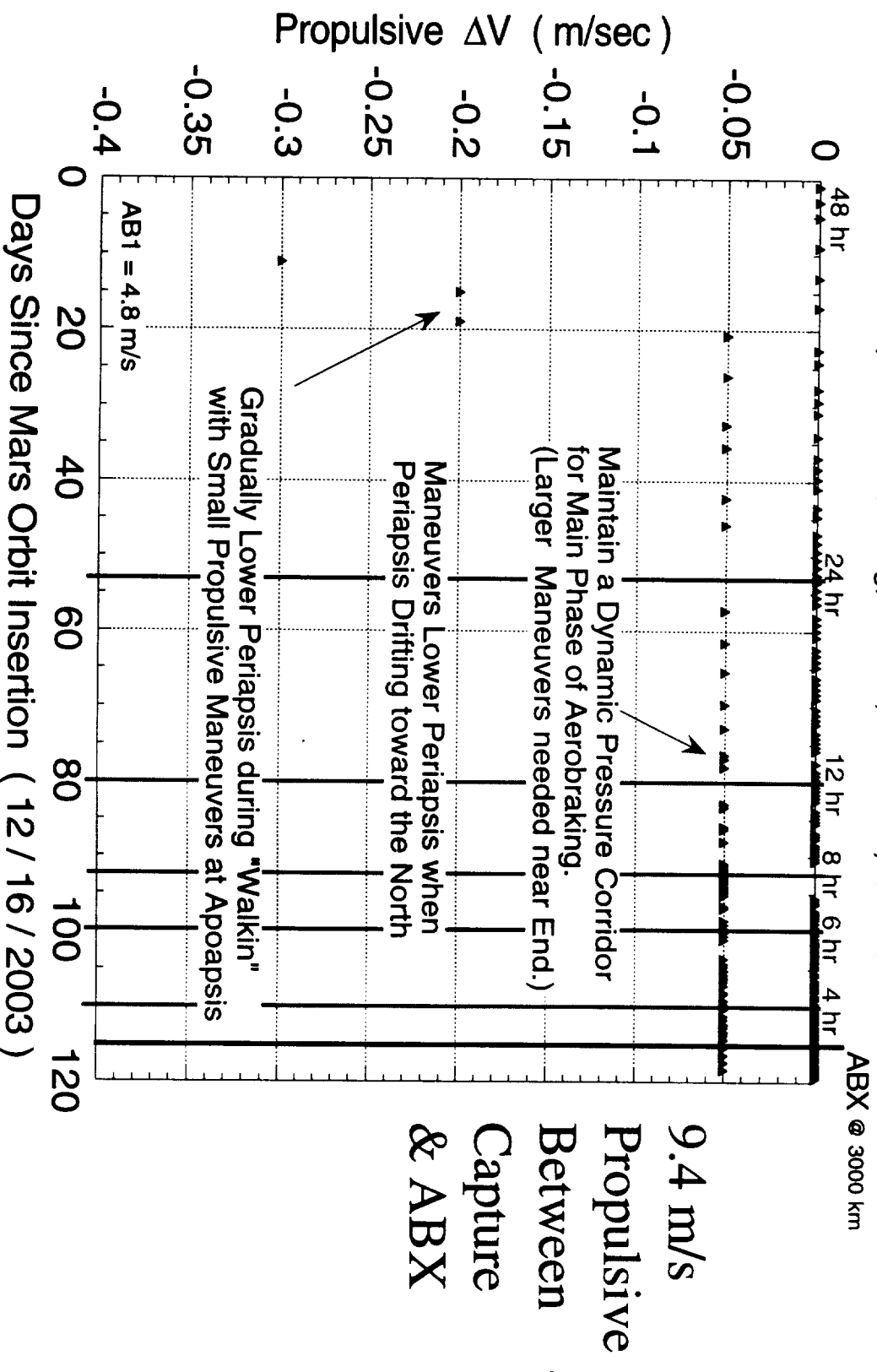
48 hr Capture Orbit, 80 kg, 2.5 m<sup>2</sup>, 0.4 N/m<sup>2</sup>, 0.2 W/cm<sup>2</sup>



# $\Delta V$ Maneuvers During Aerobraking

## DYNAMO AEROBRAKING PHASE

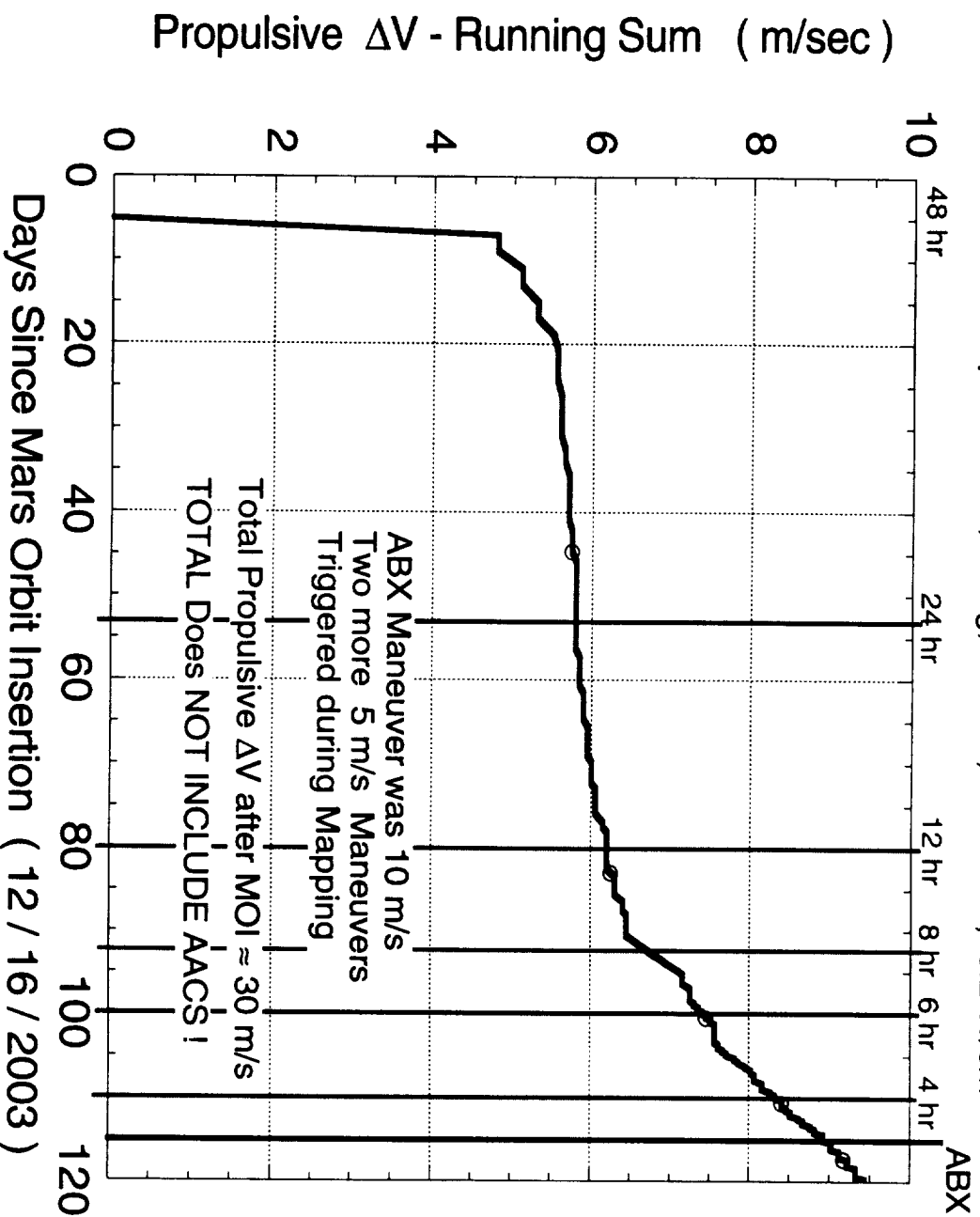
48 hr Capture Orbit, 80 kg, 2.5 m<sup>2</sup>, 0.4 N/m<sup>2</sup>, 0.2 W/cm<sup>2</sup>



# $\Delta V$ (Running Sum)

## DYNAMO AEROBRAKING PHASE

48 hr Capture Orbit, 80 kg, 2.5 m<sup>2</sup>, 0.4 N/m<sup>2</sup>, 0.2 W/cm<sup>2</sup>

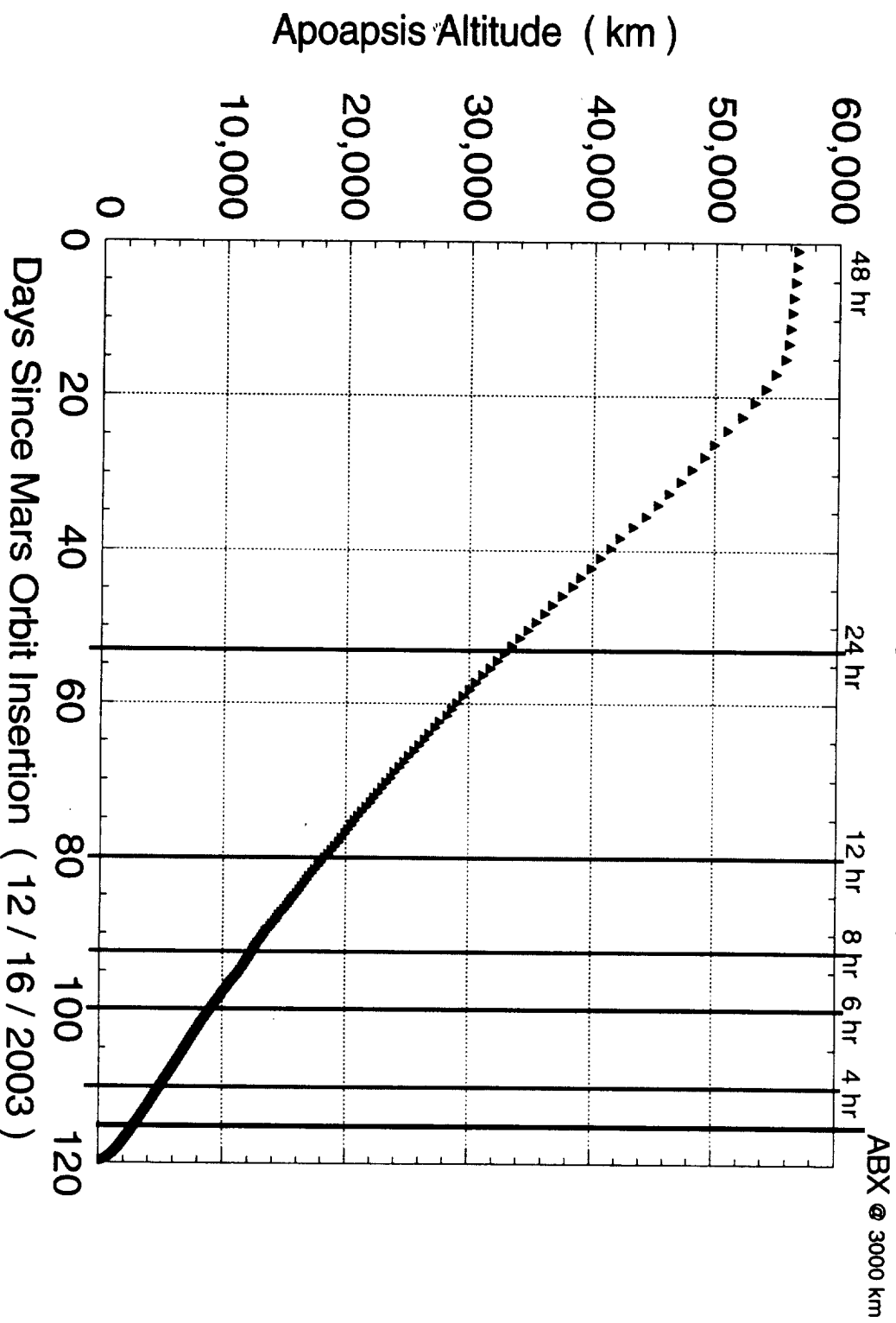


9.4 m/s  
Propulsive  
Between  
Capture  
& ABX

# APOAPSIS ALTITUDE

## DYNAMO AEROBRAKING PHASE

48 hr Capture Orbit, 80 kg, 2.5 m<sup>2</sup>, 0.4 N/m<sup>2</sup>, 0.2 W/cm<sup>2</sup>

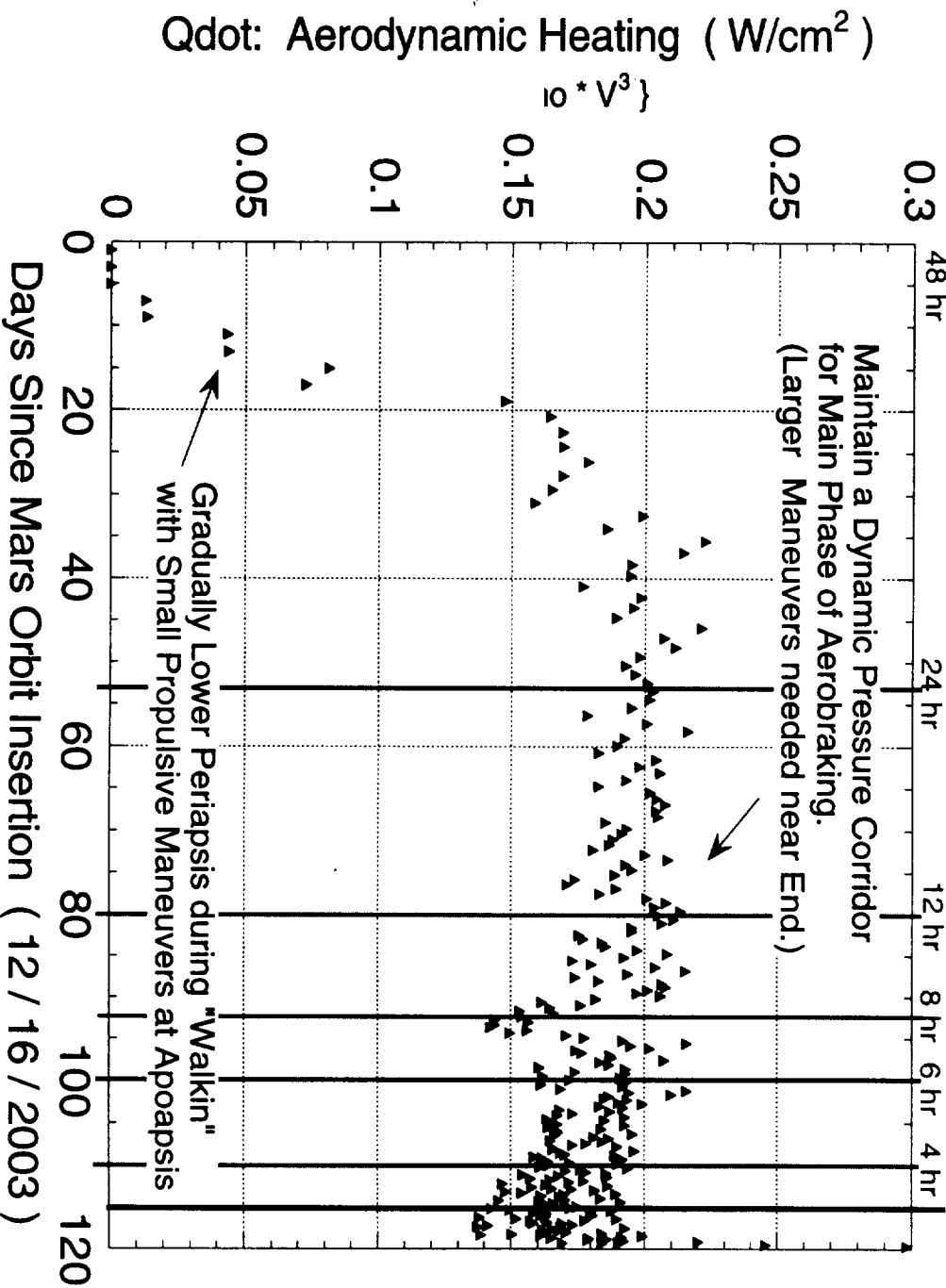


# Aerodynamic Heating Rate

## DYNAMO AEROBRAKING PHASE

48 hr Capture Orbit, 80 kg, 2.5 m<sup>2</sup>, 0.4 N/m<sup>2</sup>, 0.2 W/cm<sup>2</sup>

ABX @ 3000 km







# MAPPING PHASE

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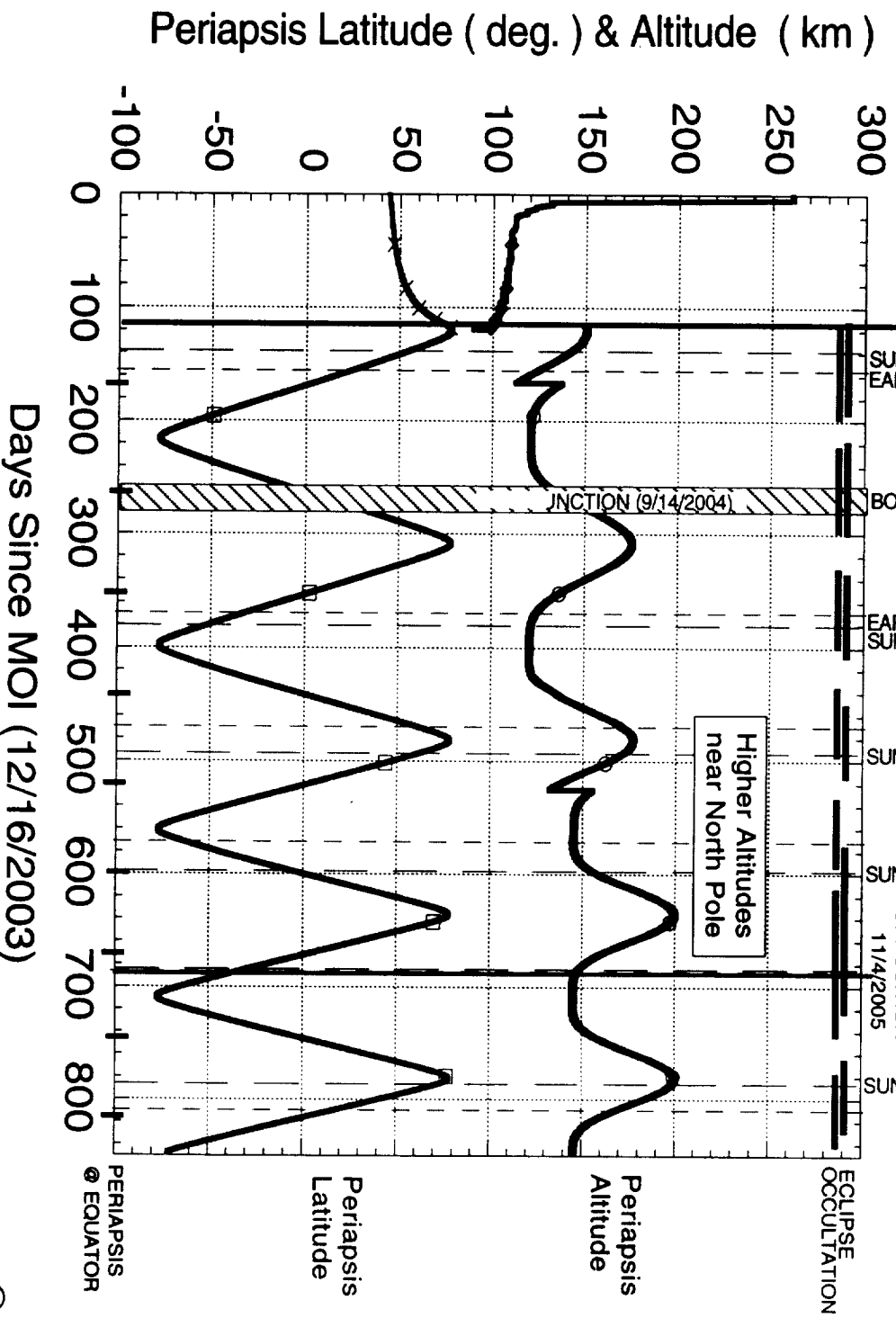
- 3000 km Apoapsis at Start of Mapping
  - To allow further Orbit Decay during Mapping with Low Altitude Periapsis
- Periapsis Precession
  - Opposite the Orbital Motion
  - Walks from “North to South to North”
  - Argument of Periapsis Precession Rate relatively constant.
  - Gives nearly Global Coverage (limited by Chosen Inclination of  $78^\circ$ )
- Conjunction puts a Gap in the Data

# PERIAPSIS ALTITUDE & LATITUDE

## DYNAMO

80 kg,  $2.5 \text{ m}^2$ ,  $0.4 \text{ N/m}^2$ ,  $0.2 \text{ W/cm}^2$

AEROBRAKING | MAPPING

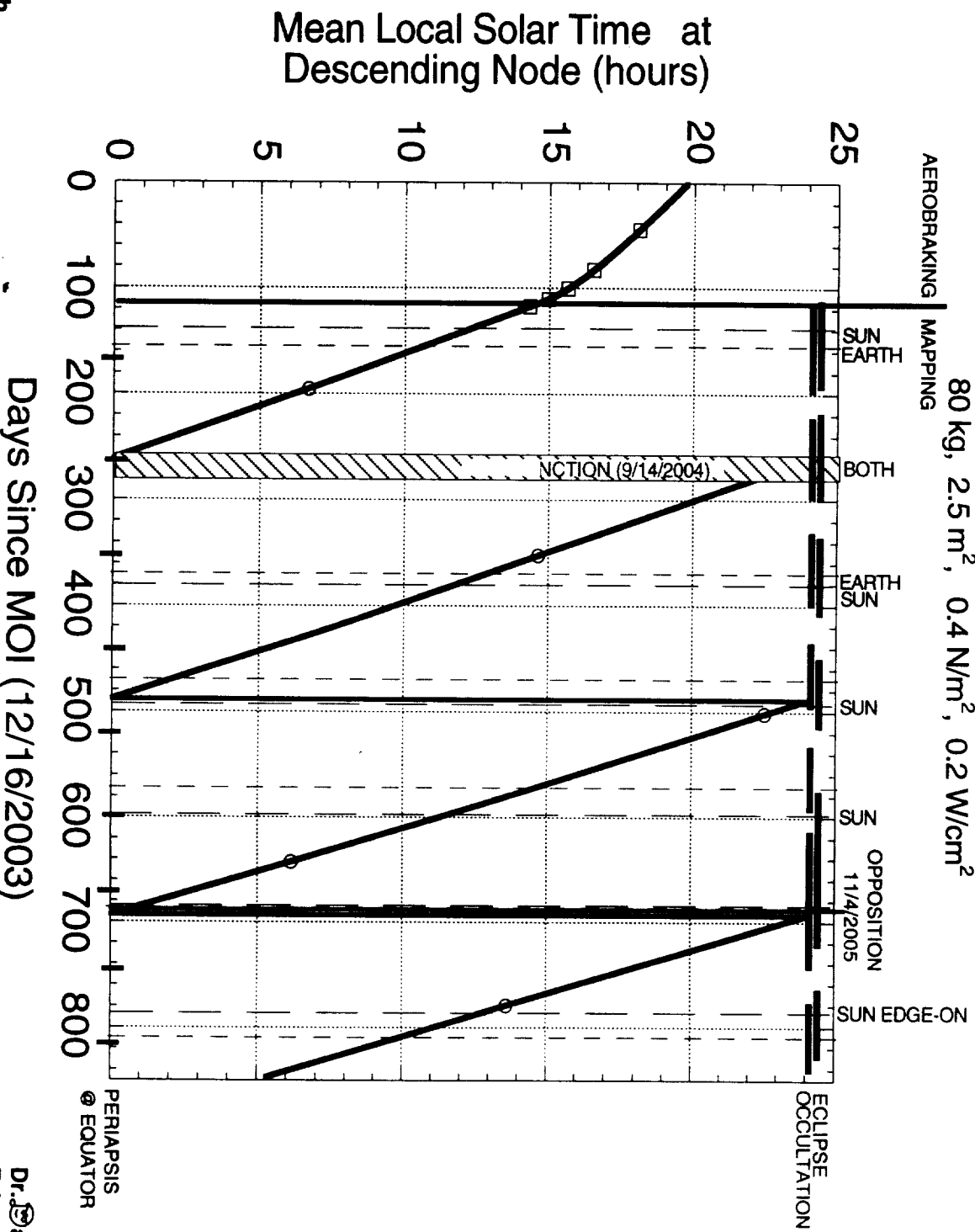


# LOCAL SOLAR TIME at Descending Node

## DYNAMO

80 kg, 2.5 m<sup>2</sup>, 0.4 N/m<sup>2</sup>, 0.2 W/cm<sup>2</sup>

AEROBRAKING MAPPING



1. *Journal of Management Studies*, 1990, 27, 1, 1-14.

80 kg, 2.5 m<sup>2</sup>, 0.4 N/m<sup>2</sup>, 0.2 W/cm<sup>2</sup>

## AEROBRAKING





## SUMMARY

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- A Possible Dynamo Orbit Strawman was Shown
- Maximum Eclipse = 50 min.
  - 60 min. was “Dangerous” for MGS
- Average Dynamic Pressure  $\approx 0.4 \text{ N/m}^2$ 
  - Original MGS used Average  $\approx 0.6 \text{ N/m}^2$  (100% Heating Margin)
  - Current MGS (with Broken Wing) uses Average  $\approx 0.2$  to  $0.3 \text{ N/m}^2$
  - Does Not Protect against Dust Storm Induced Increases !!
- Provides nearly Global Sampling at Periapsis